

**1-Body Motion**  $\vec{f} = \dot{\vec{p}} = m\ddot{\vec{r}} \quad \vec{p} = m\vec{v} = m\dot{\vec{r}}$

$\vec{l} \equiv \vec{r} \times \vec{p} \quad \vec{\tau} \equiv \vec{r} \times \vec{f} = \dot{\vec{l}} \quad \vec{f}_{EM} = q(\vec{E} + \vec{v} \times \vec{B})$

$\vec{f}_{air} = -(bv + cv^2)\hat{v} = \vec{f}_{lin} + \vec{f}_{quad} \quad \text{Reynolds } R \equiv Dv\rho / \eta$

sphere, diameter D:  $b = \beta D \quad c = \gamma D^2 \quad R = 48 f_{quad} / f_{lin}$

**Rocket Motion**  $m\dot{v} = -\dot{m}v_{ex} + F^{EXT}$

**Miscellaneous**  $1 \text{ m/s} \approx 2 \text{ mph}$

$f_{friction} = \mu F_N \quad \vec{f}_{grav} = -\frac{GMm}{r^2}\hat{r}$

**Collective Motion** \* assuming Newton's 3<sup>rd</sup> Law  $\rightarrow F^{INT}$  cancel

**Notation** for collective properties

- unsubscripted capital letter  $\rightarrow$  "TOTAL", except for ...
- unsubscripted capital position, velocity, accel  $\rightarrow$  "OF THE CM"
- ◇ subscript  $\neq$  coordinate index  $\rightarrow$  "OF"
- ◆ no superscript  $\rightarrow$  "RELATIVE TO ORIGIN"
- ◆ superscript ()  $\rightarrow$  "RELATIVE TO (POINT)"
- ◆ superscript prime'  $\rightarrow$  "RELATIVE TO THE CM"

**CM:**  $M\vec{R} \equiv \sum_i m_i \vec{r}_i \quad \vec{P} = M\dot{\vec{R}} \quad \vec{L}_{CM} = \vec{R} \times \vec{P}$

**Rotating Body:** for any BODY-FIXED vector  $\vec{B}$ ,  $\dot{\vec{B}} = \vec{\omega} \times \vec{B}$

**Moment of Inertia:** for any BODY-FIXED point B,

$I_{\hat{\omega}}^{(B)} \equiv \sum m_i |\vec{r}_i^{(B)} \times \hat{\omega}|^2 \quad L_{\omega}^{(B)} = I_{\hat{\omega}}^{(B)} \omega \quad T^{(B)} = \frac{1}{2} I_{\hat{\omega}}^{(B)} \omega^2$

**Uniform Gravity:** If  $\vec{f}^{EXT} = m\vec{g}$  only  $\rightarrow \vec{F}^{EXT} = M\vec{g} \quad \vec{\tau}^{EXT} = \vec{R} \times M\vec{g} \quad \vec{\tau}'^{EXT} = 0 \quad U^{EXT} = MgH$

**Variational Calc / Mech** \* Gen. coord  $q_i$  must be indep

$S \equiv \int_{t_1, \vec{q}_1}^{t_1, \vec{q}_1} dt L(q_i, \dot{q}_i, t) \quad \delta S = 0 \rightarrow \frac{\partial L}{\partial q_i} = \frac{d}{dt} \left( \frac{\partial L}{\partial \dot{q}_i} \right)$  for each  $q_i$

$H \equiv \dot{q}_i (\partial L / \partial \dot{q}_i) - L$  conserved when  $\partial L / \partial t = 0$

**Mechanics** Principle of Least Action :

$L = T - U \rightarrow \delta S = 0 @ \text{ true } \{q_i(t)\}$

Gen. force  $Q_i \equiv \frac{\partial L}{\partial q_i}$ , momentum  $p_i \equiv \frac{\partial L}{\partial \dot{q}_i}$

$H \equiv p_i \dot{q}_i - L$  equals  $T+U$  when  $\vec{r}_a = \vec{r}_a(q_i)$

**EOM in Inertial Frames**

$\vec{F}^{EXT} = \dot{\vec{P}}$

$\vec{\tau}^{EXT, (A)} = \dot{\vec{L}}^{(A)}$  if reference point A

- is the CM, or
- is not accelerating, or
- $\ddot{\vec{r}}_A \parallel (\vec{R} - \vec{r}_A)$

$T + U^{EXT} + U^{INT} = \text{conserved for conservative forces}$

**Decompositions**

$\vec{P} = \vec{P}_{CM} \quad I_{\hat{\omega}}^{(B)} = I_{CM, \hat{\omega}}^{(B)} + I'_{\hat{\omega}}$

$\vec{L} = \vec{L}_{CM} + \vec{L}' \quad \vec{\tau}_{CM}^{EXT} = \dot{\vec{L}}_{CM}$

$T = T_{CM} + T' \quad T = T^{(\text{stationary point})}$