



Chapter 2: Strain

Chapter Objectives

- ✓ Understand the concepts of normal and shear strain
- ✓ Apply the concept to determine the strain for various types of problems

Strain is a measure of geometric deformation.

Critical
become proficient
in symbolic algebra

key: understand dimensional analysis

$$[\sigma] = \frac{\text{force}}{\text{area}} = \text{stress}$$

$$[\tau] = \frac{\text{force}}{\text{area}}$$

$$[A] = \text{area}$$

$$A = \frac{\pi}{4} d^2 \quad [d] = \text{length}$$

$$\left[\frac{\pi}{4}\right] = 1 \quad [d^2] = \text{area} = \text{length}^2$$

$$\sigma = E \cdot \epsilon$$

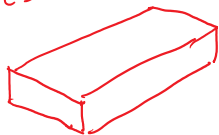
↑
Young's
Modulus

$$[E] = \text{stress} \quad [\epsilon] = 1 = \text{length}^0 = 1$$

Dimensions must match
on both sides of any
equation. $\Rightarrow [\sigma] = [E \cdot \epsilon] = [E] \cdot [\epsilon]$

Question

If a rectangular
bar of some metal
is heated uniformly,
does its shape change?



\Rightarrow presence of normal
strain, but no
shear strain

If a ring
is held upright
and stepped on,
~~it~~ will its
shape change?



\Rightarrow presence of
shear strain
(and normal
strain)

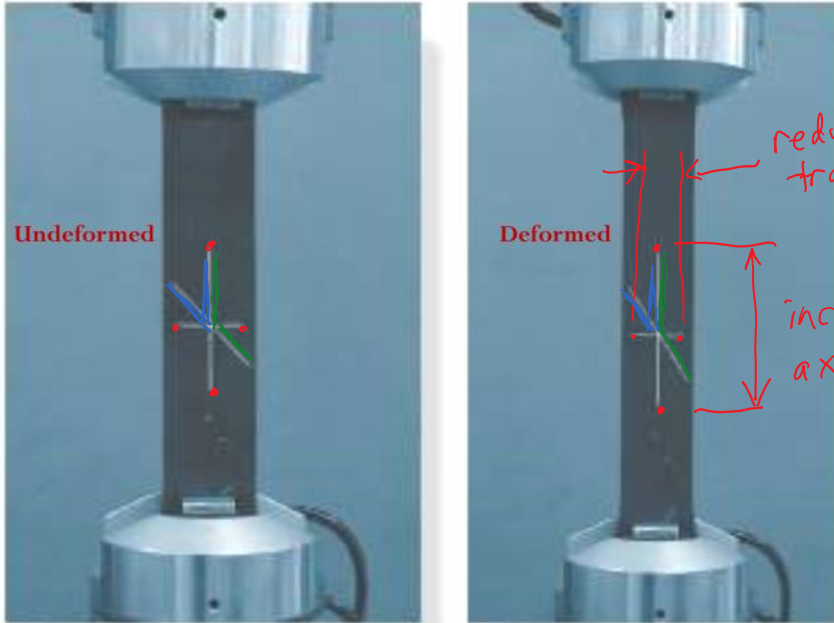
due to normal strain

due to shear strain

DEFORMATION: change in length or shape of a body when forces are applied (or change in temperature)



DEFORMATION: change in length or shape of a body when forces are applied (or change in temperature)



Rubber membrane subject to tension

Rectangle of one aspect ratio to
a rectangle of a different
aspect ratio

Extensional strain (normal strain)

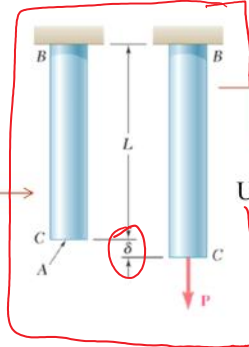
Change in length of a member divided by its original length (i.e., deformation per unit length)

$$\epsilon = \frac{\delta}{L} = \frac{L_{final} - L_{initial}}{L_{initial}}$$

change in length

$$[\epsilon] = 1$$

Undeformed configuration



Deformed configuration

Uniform strain along member AB

Strain is dimensionless!

Recall point-wise definition of stress:

$$\sigma = \lim_{\Delta A \rightarrow 0} \frac{\Delta F}{\Delta A}$$

infinitesimal increase in length at x

Similarly, we have a point-wise definition of strain:


$$\epsilon = \lim_{\Delta x \rightarrow 0} \frac{\Delta \delta}{\Delta x} = \frac{d\delta}{dx}$$

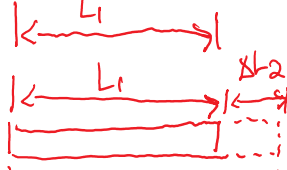
True vs Engineering Strain

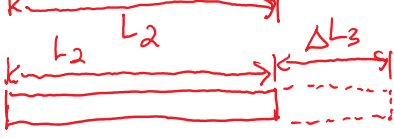
We just defined "engineering strain", $\epsilon = \frac{\delta}{L_i}$

↖ change in length
↖ initial length

"True strain" accounts for change in length of the bar as strain increases

1  $\Delta \epsilon_1 = \frac{\Delta L_1}{L_0}$

2  $\Delta \epsilon_2 = \frac{\Delta L_2}{L_1}$

3  $\Delta \epsilon_3 = \frac{\Delta L_3}{L_2}$

etc

$$\epsilon_{true} = \int_0^{\epsilon_{final}} d\epsilon$$

$$= \int_{L_0}^{L_f} \frac{dL}{L}$$

$$= \ln\left(\frac{L_f}{L_0}\right)$$

$$= \ln\left(\frac{L_0 + \delta}{L_0}\right)$$

$$= \ln\left(1 + \frac{\delta}{L_0}\right)$$

$[d\epsilon] = 1$

$\left[\frac{dL}{L}\right] = 1$

For most practical engineering purposes, (or "structural") $\epsilon_{true} = \ln\left(1 + \epsilon_{engineering}\right)$

the true strain is very, very, very small.

For those cases $\ln(1 + \epsilon_{eng.}) \approx \epsilon_{eng.}$
 $\epsilon_{true} \approx \epsilon_{eng.}$

Taylor expansion

$$\ln(1 + \epsilon_{eng}) = \epsilon_{eng} - \frac{1}{2} \epsilon_{eng}^2 + \frac{1}{3} \epsilon_{eng}^3 - \frac{1}{4} \epsilon_{eng}^4 + \dots$$

when $\epsilon_{eng} \ll 1$ (much smaller than 1), $\epsilon_{eng}^2, \epsilon_{eng}^3, \text{etc.} = 0$
 (get rid of higher orders of ϵ_{eng})

$$\ln(1 + \epsilon_{eng}) = \epsilon_{eng} \text{ for small } \epsilon_{eng}$$

True vs Engineering Strain

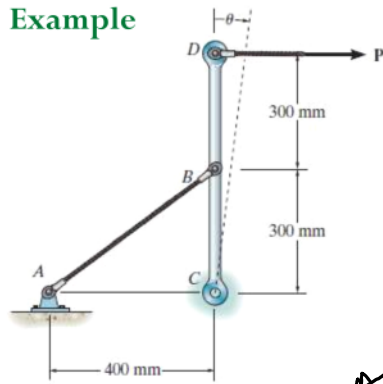
For $L_i = 10''$

δ	$\epsilon_{eng} = \frac{\delta}{L_i}$	$\epsilon_{true} = \ln\left(\frac{L_f}{L_i}\right)$	Error
0.01''	0.001	0.00099	0.05%
0.05''	0.005	0.00498	0.25%
0.1''	0.01	0.00995	0.5%
1''	0.1	0.0953	4.9%
5''	0.5	0.4054	23.3%

clearly
acceptable
for TAM251
analysis
borderline

usually in engineering
5% error and less is
acceptable

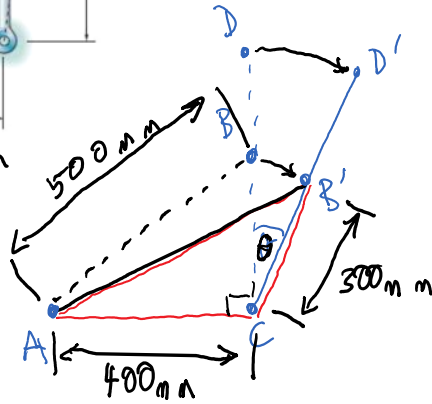
Example



Part of a control linkage of an airplane consists of a rigid member CDB and a flexible cable AB. If a force is applied at the end D of the member and causes a normal strain in the cable of 0.0035 mm/mm, determine the displacement of point D. Originally the cable is unstretched.

Method 1: Trigonometry

$$L_{AB} = 500 \text{ mm}$$



$$\epsilon_{AB} = 0.0035 = \frac{L_{AB'} - L_{AB}}{L_{AB}}$$

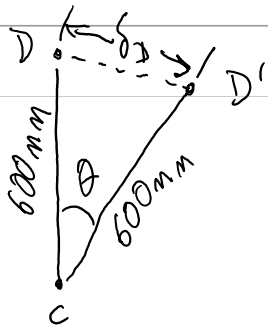
$$L_{AB'} = L_{AB}(1 + 0.0035) = 501.75 \text{ mm}$$

law of cosines to red triangle

$$(501.75 \text{ mm})^2 = (300 \text{ mm})^2 + (400 \text{ mm})^2$$

$$-2(300 \text{ mm})(400 \text{ mm})\cos(90^\circ + \theta)$$

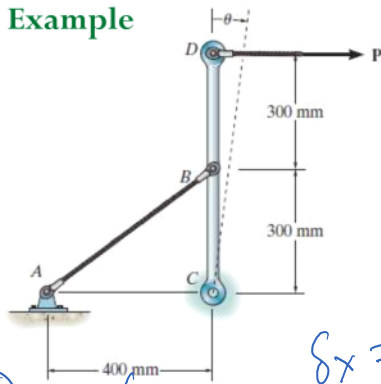
$$\therefore \theta = \cos^{-1}(\text{mm}) - 90^\circ \approx 0.419^\circ = 0.0073 \text{ rad}$$



$$\delta_D^2 = (600 \text{ mm})^2 + (600 \text{ mm})^2 - 2(600 \text{ mm})^2 \cdot \cos(0.0073 \text{ rad})$$

$$\delta_D = 4.38 \text{ mm}$$

Example



Part of a control linkage of an airplane consists of a rigid member CDB and a flexible cable AB. If a force is applied at the end D of the member and causes a normal strain in the cable of 0.0035 mm/mm, determine the displacement of point D. Originally the cable is unstretched.

Method 2: Assume rotations are small

①

$$\delta_x = L \cdot \sin \theta$$

$$\delta_y = L - L \cos \theta$$

For small θ

$$\sin \theta \approx \theta$$

$$\cos \theta \approx 1$$

$$\tan \theta \approx \theta$$

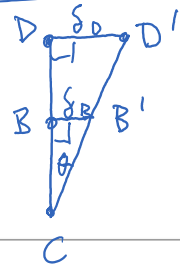
$$\delta_x \approx L \cdot \theta \quad \delta_y \approx 0$$

②

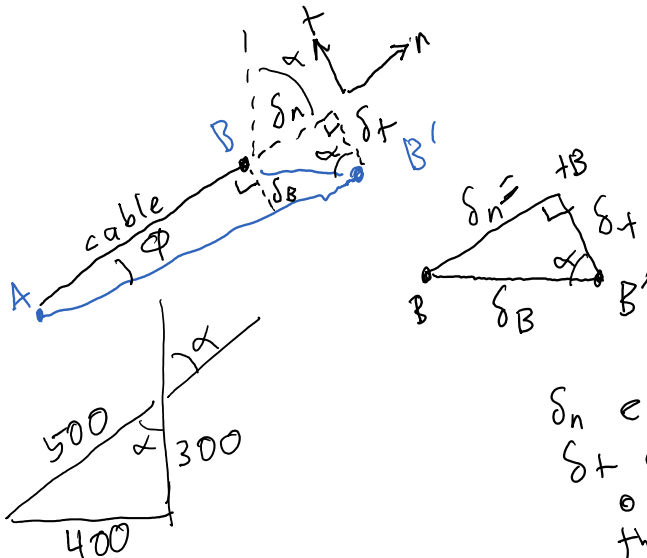
$$\frac{\delta_{AB}}{L_{AB}} = 0.0035$$

$$\delta_{AB} \approx 1.75 \text{ mm}$$

③



Triangle CDD' is similar with CBB'



$$\frac{\delta_D}{600 \text{ mm}} = \frac{\delta_B}{300 \text{ mm}}$$

$$\Rightarrow \delta_D = 2 \cdot \delta_B$$

δ_n extension of cable = δ_{AB}
 δ_t due to rotation of the cable through small angle ϕ

$$\delta_{AB} = \delta_B \cdot \sin \alpha$$

$$\delta_B = \frac{\delta_{AB}}{\sin \alpha}$$

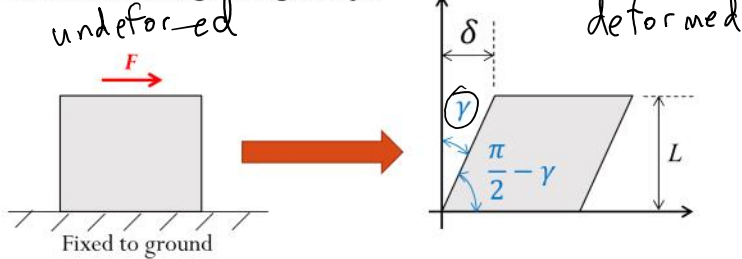
$$\delta_D = 2 \cdot \delta_B = \frac{2 \cdot \delta_{AB}}{(4/5)} = \frac{2 \cdot (1.75 \text{ mm})}{0.8}$$

$$\delta_D = 4.375 \text{ mm}$$

Shear Strain

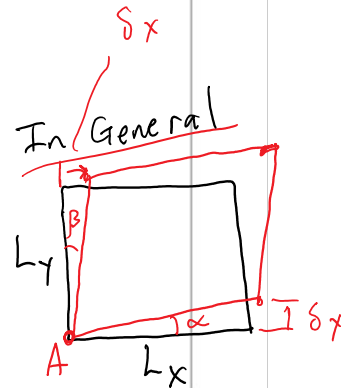
Axial loads: change in length

Shear loads: change in angle/shape



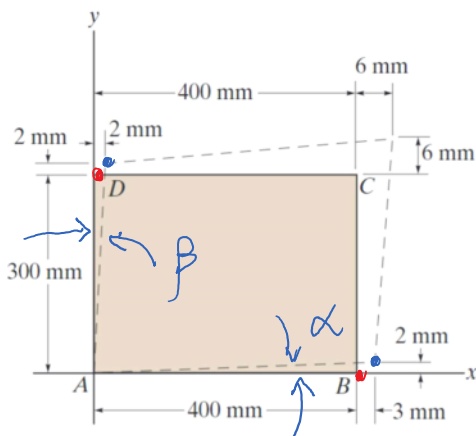
Shear strain = Change in angle that was originally at 90 degrees ($\frac{\pi}{2}$)
 $= \gamma$ (for now, we consider shear strain **magnitudes** only)

$$\tan \gamma = \frac{\delta}{L} \approx \gamma \text{ for small } \gamma$$



$$\gamma_A = \alpha + \beta = \frac{\delta_y}{L_x} + \frac{\delta_x}{L_y}$$

Example



The rectangular plate is deformed into the shape shown by the dashed lines.

Determine

- the average normal strain along diagonal BD
- the average shear strain at corner B

$$a) \epsilon_{BD} = \frac{L_{B'D'} - L_{BD}}{L_{BD}}$$

$$L_{BD} = 500 \text{ mm}$$

$$B' = (403, 2) \text{ mm}$$

$$D' = (2, 302) \text{ mm}$$

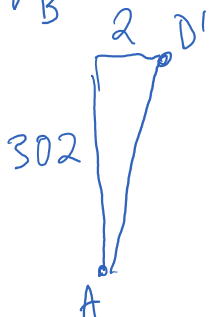
$$L_{B'D'} = \sqrt{(403-2)^2 + (2-302)^2} \text{ mm}$$

$$= 500.8 \text{ mm}$$

$$\epsilon_{BD} = \frac{500.8 - 500}{500} = 1.6 \times 10^{-3}$$



$\gamma_B = \alpha + \beta$



$\alpha = \tan^{-1}\left(\frac{2}{403}\right) = 0.0050$

$\beta = \tan^{-1}\left(\frac{2}{302}\right) = 0.0066$

$\Rightarrow \boxed{\gamma_B = 0.0050 + 0.0066 = 0.0116 \text{ rad}}$

Measurement of Strain

- **Direct measurement:**

- Initial and final lengths of some section of the specimen are measured, perhaps by some handheld device such as a ruler
- Axial strain computed directly by following formula:

$$\epsilon = \frac{\delta}{L} = \frac{L_{final} - L_{initial}}{L_{initial}}$$

- Accurate measurements of strain in this way may require a fairly large initial length

Measurement of Strain

CEE 300 ME 330
TAM 324

• Contact Extensometer:

- A clip-on device that can measure very small deformations
- Two clips attach to a specimen before testing
- The clips are attached to a transducer body

$$\epsilon = \frac{\delta}{L} = \frac{L_{final} - L_{initial}}{L_{initial}}$$

- The transducer outputs a voltage
- Changes in voltage output are converted to strain



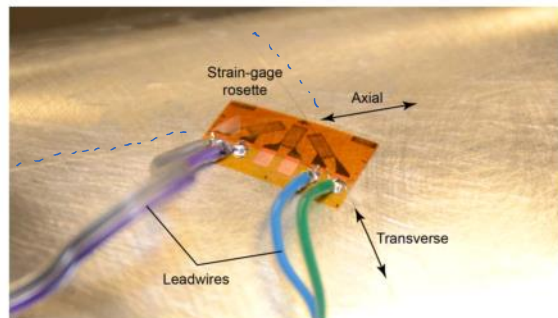
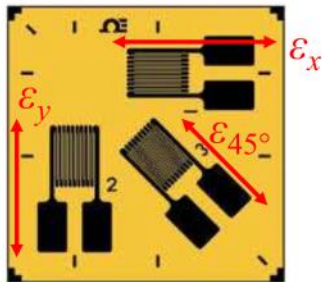
High-temperature contact extensometer. From instron.com



A tensile test in the Materials Testing Instructional Laboratory, Talbot Lab, UIUC

Measurement of Strain

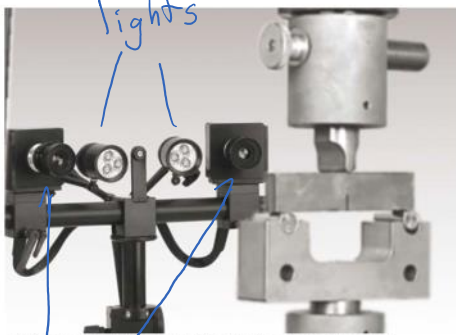
- Strain gages
 - Small electrical resistors whose resistance changes with strain
 - Change in resistance can be converted to strain measurement
 - Often sold as “rosettes,” which can measure normal strain in two or more directions
 - Can be bonded to test specimen



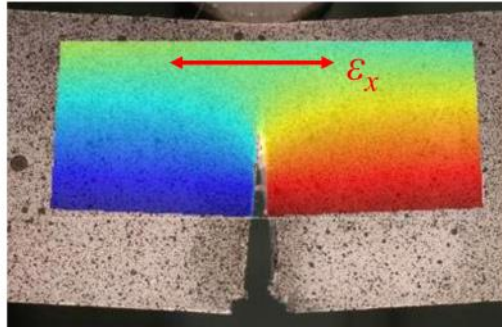
Measurement of Strain

- **Digital Image Correlation (DIC)**

- Image placed on surface of test specimen
- Image may consist of speckles or some regular pattern
- Deformation of image tracked by digital camera
- Image analysis used to determine multiple strain components



DIC system analyzing a notch fracture test, from trillion.com



Strain field in a notch fracture test, as measured using DIC.
From barthelat-lab.mcgill.ca

2 cameras
at an angle