

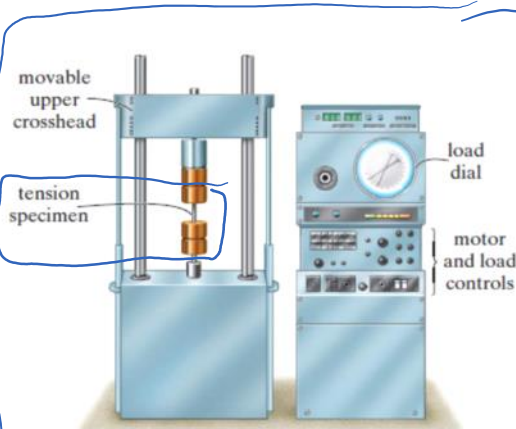


## Chapter 3: Mechanical Properties of Materials

### Chapter Objectives

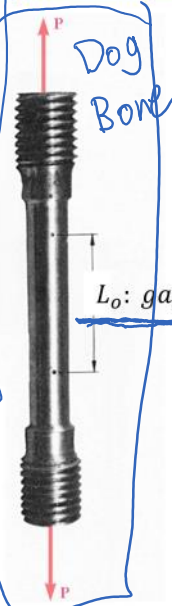
- ✓ Understand how to measure the stress and strain through experiments
- ✓ Correlate the behavior of some engineering materials to the stress-strain diagram

# Tension and compression test



<https://www.youtube.com/watch?v=AFCAS1ocFgo>

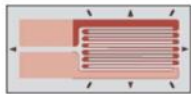
<https://www.youtube.com/watch?v=jbslR-6GVCs>



universal testing machine

~ Instron

~ CEE 300, TAM 324  
ME 330

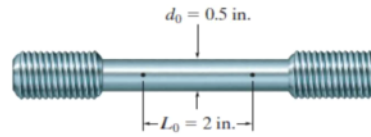


Electrical-resistance strain gauge



Typical steel specimen with attached strain gauge.

# Stress-strain diagram



## Uniaxial tension test:

- Specimen is stretched at a very slow, constant rate
- Measure distance  $L$  and load  $P$  at frequent intervals
- Convert  $L$  and  $P$  to **engineering** stress-strain data

$$\epsilon = \frac{L - L_0}{L_0}$$

axis of the specimen

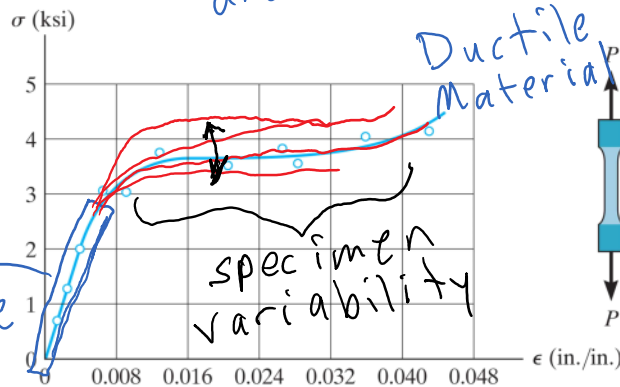
$$\sigma = \frac{P}{A_0}$$

initial cross-section area of the specimen

Note that two stress-strain diagrams for a particular material will be similar, but not identical, due to variability in:

- Material composition
- Specimen imperfections
- Rate of loading
- Ambient temperature

$y = m \cdot x$   
type response



can find many material properties

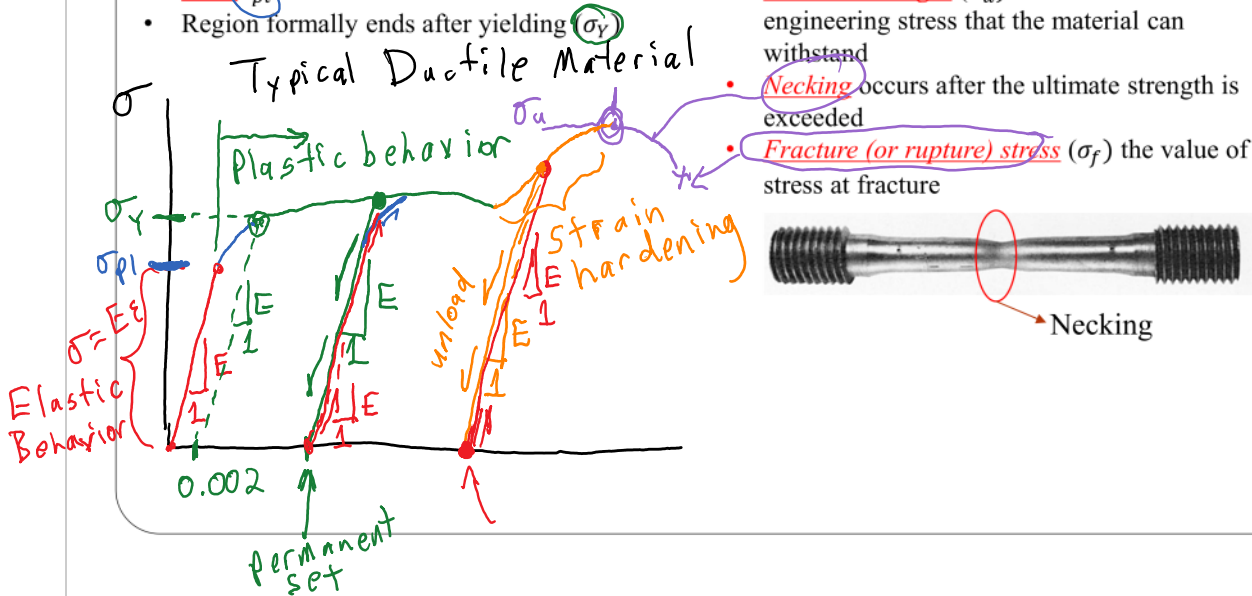
## Characteristics of stress-strain diagram

### Elastic behavior (recoverable deformation):

- Reversible Deformation
- Materials loaded in this region regain its original shape if load is removed
- The curve is a straight line throughout most of this region (stress is proportional to strain).
- Linear region is limited by the proportional limit ( $\sigma_{pl}$ )
- Region formally ends after yielding ( $\sigma_y$ )

### Plastic behavior (permanent deformation):

- Stresses above the elastic limit will cause the material to deform permanently
- Yield strength ( $\sigma_y$ ): highest stress that the material can withstand without undergoing significant yielding
- Ultimate strength ( $\sigma_u$ ) is the maximum value of engineering stress that the material can withstand
- Necking occurs after the ultimate strength is exceeded
- Fracture (or rupture) stress ( $\sigma_f$ ) the value of stress at fracture



$$[E] = [\sigma] = \frac{\text{force}}{\text{area}}$$

Linearly elastic regime

$$\text{slope} = E : \sigma = E \cdot \epsilon \quad [E] = 1$$

$\sigma_y$  = yield stress - found by 0.2% offset method } "flat" region  
 - unloadings after yielding } after we reach  $\sigma_y$  is  
 also have slope = E } known as "yielding"

Upon reloading, we observe slope of E  
 In strain hardening,  $\sigma_y$  increases!

### Necking

- rapid, localized reduction of cross-section
- locally,  $\sigma$  increases rapidly

- Final behavior we observe in a ductile material prior to rupture (fracture)

## Ductile materials

- Rupture occurs along a cone-shaped surface that forms an angle of approximately  $45^\circ$  with the original surface of the specimen

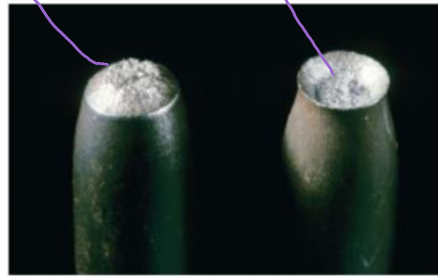
- Shear is primarily responsible for failure in ductile materials

- Axial loading: maximum shear stress occurs at  $45^\circ$

"shear lips"  $\Rightarrow$  happen where  $\tau$  is maximized

cone

cup



- Ways to specify ductility:

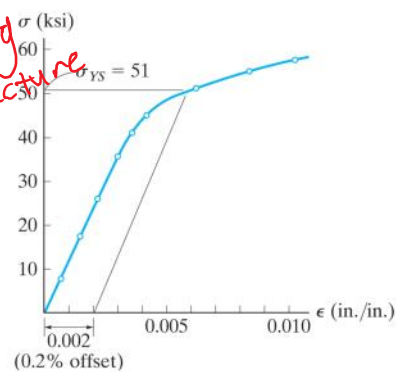
- $\% \text{ elongation} = 100 \times \frac{(L_f - L_0)}{L_0}$

- $\% \text{ area reduction} = 100 \times \frac{(A_0 - A_f)}{A_0}$

at the neck

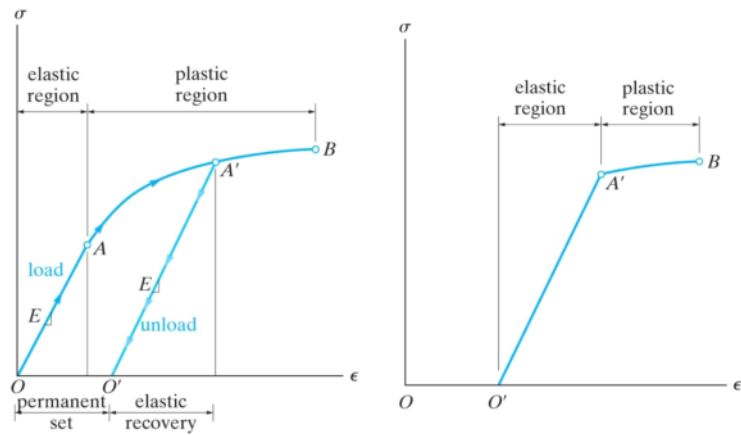
- The yield strength  $\sigma_Y$  is often defined as the stress at which unloading would produce a plastic or permanent strain of 0.002 (offset method)

after testing to fracture

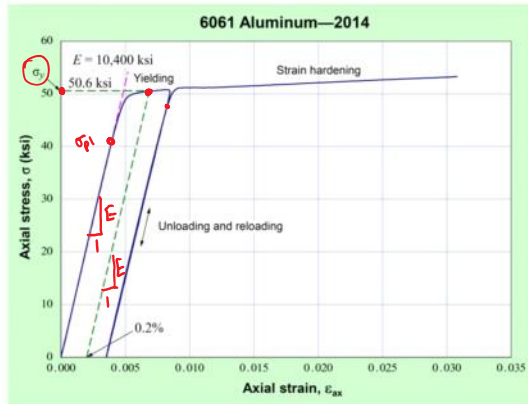


**Strain-Hardening:**

- If a specimen of a ductile material is loaded into the plastic region and then unloaded, *elastic strain* is recovered as the material return to its equilibrium state
- The *plastic strain* remains, resulting in *permanent deformation*
- If material is reloaded, it will have a higher yielding point – *strain hardening*. The material has a **greater elastic region**, however it has **reduced ductility**



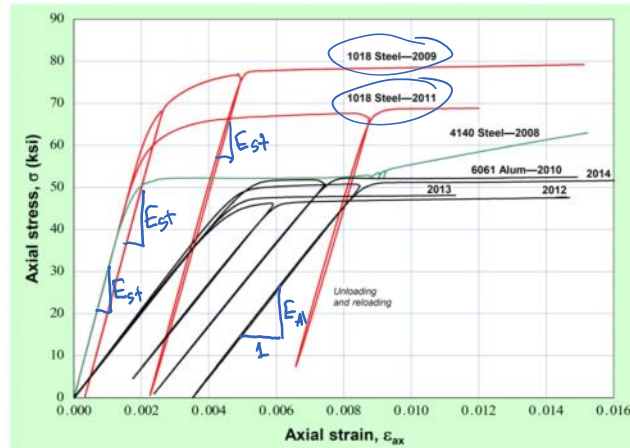
## Some Example Data: Loading and unloading of some structural metals



$E$  is determined by  
the primary element  
of the alloy

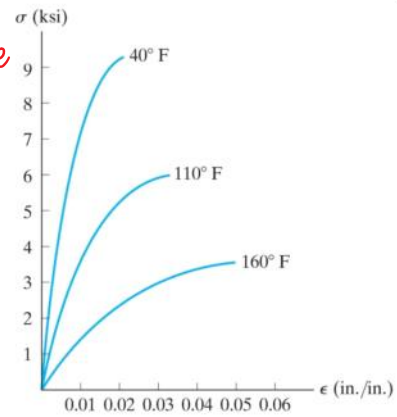
Steels have one  $E$  value

Aluminums have one different  $E$  value  
 $E_{tc}$ .



## Brittle materials

- Material that exhibit little or no yielding before failure
- Absence of necking
- Rupture occurs along a surface perpendicular to the load
- Normal stress is primarily responsible for failure of brittle materials



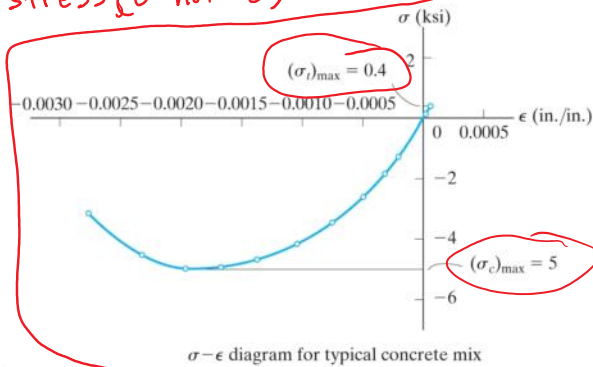
↑ brittle materials fail due to max normal stress ( $\sigma$  not  $\tau$ )

## Concrete (brittle material)

- Maximum compressive strength is substantially larger than the maximum tensile strength
- For this reason, concrete is almost always reinforced with steel bars or rods whenever it is designed to support tensile loads



Compression causes material to bulge out



$\sigma$ - $\epsilon$  diagram for typical concrete mix

## Hooke's law

For small deformations (elastic region up to yield strength)

$$\sigma = E \epsilon$$

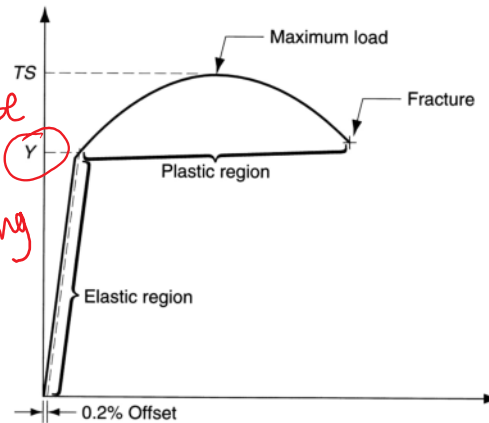
We will use  
this all  
semester long

## For plastic deformation:

For the plastic region:  $\sigma = K \epsilon^n$  } power law curve fit

K = strength coefficient

n = strain hardening exponent

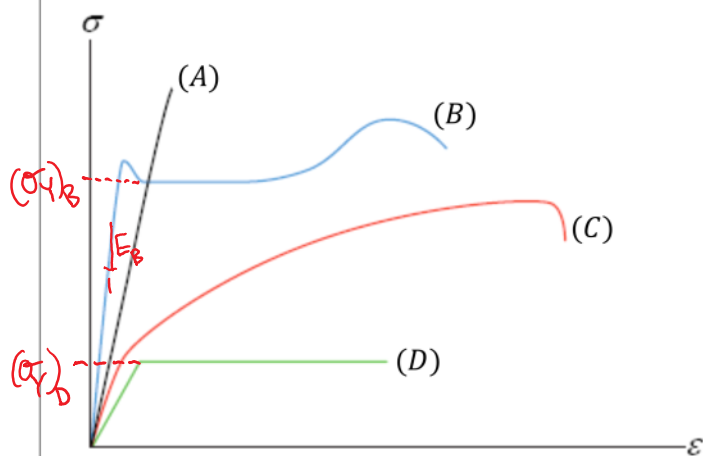


	$\sigma_u$	$\sigma_Y$	E
Steel	400 MPa	250 MPa	200 GPa
Al	110 MPa	95 MPa	70 GPa
Concrete (compression)	28 MPa	-	25 GPa

depends on alloy and pre-working of the material

very reliable

determined by the 0.2% offset method



Which material has highest stiffness  
(Young's modulus,  $E$ )?

*B, steepest linear region*

Which material is most ductile?

*C, highest  $\epsilon$  before rupture*

Which material is most brittle?

*A, least yielding region (none)*

Which material has the lowest yield strength  $\sigma_Y$ ?

*D*

# Hooke's law for shear stress and strain

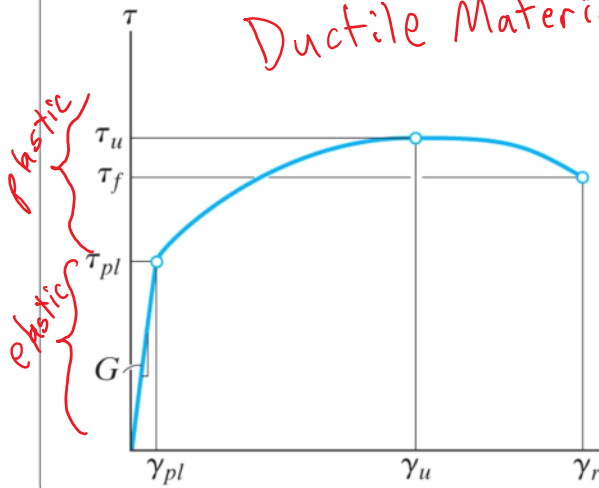
$$\tau_{xy} = \boxed{G} \gamma_{xy}$$

Hookers law  
-linear relation  
between  $\tau$  and  $\gamma$

usually determined  
using torsion  
experiments

Modulus of rigidity or shear modulus

Ductile Material



ONLY TWO OF THE THREE  
MATERIAL CONSTANTS ARE  
INDEPENDENT IN ISOTROPIC  
MATERIALS!

shear  
modulus

$$G = \frac{E}{2(1 + \nu)}$$

$$[G] = \frac{\text{force}}{\text{area}}$$

$$[E] = \frac{\text{force}}{\text{area}}$$

$$[\nu] = 1$$

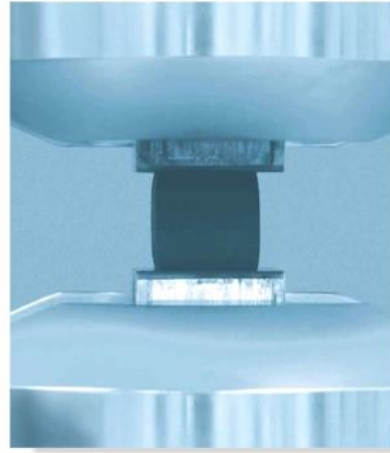
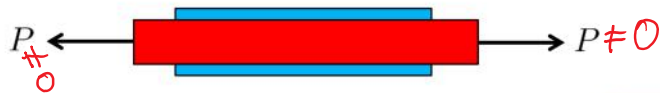
Poisson's Ratio

## Poisson's ratio

Undeformed configuration



Deformed configuration



Axial (normal) strain  $\epsilon_x = \frac{\delta}{L}$

Poisson's ratio:  $\nu = -\frac{\text{lateral strain}}{\text{axial strain}}$

Lateral strain  
 $\epsilon_z = \epsilon_y = -\nu \epsilon_x$   
 Poisson's ratio  $\nu$   
 axial strain

$-1 < \nu < 0.5$

some materials do have  $\nu < 0$

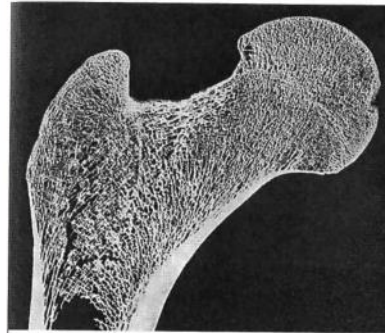
$\nu \approx 0.3$  for metals

Axial extension  $\rightarrow$  Lateral contraction

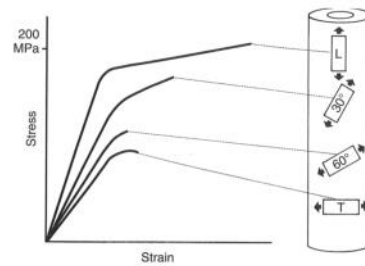
$\nu = 0.5 \Rightarrow$  isochoric material  $\Rightarrow$  "volume-preserving"

### Isotropic vs. anisotropic materials:

- Isotropic: material properties are independent of the direction
- Anisotropic: material properties depend on the direction



↑ metals  
tend to  
be isotropic  
↓



# Repeated Loadings – Fatigue

- If stress does not exceed the *elastic limit*, the specimen returns to its original configuration
- However, this is not the case if the loading is repeated thousands or millions of times
- In such cases, rupture will happen at a stress lower than the *fracture stress* - this phenomenon is known as fatigue

